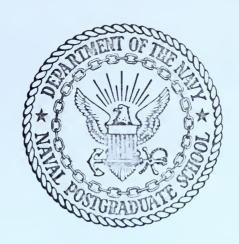
OF A LOWER CONFIDENCE LIMIT ESTIMATE FOR SERIES SYSTEM RELIABILITY

Samuel Hall Evans



NAVAL POSTGRADUATE SCHOOL

Monterey, California



THESIS

EVALUATION OF THE ACCURACY

OF A LOWER CONFIDENCE LIMIT ESTIMATE

FOR SERIES SYSTEM RELIABILITY

by

Samuel Hall Evans

Thesis Advisor:

W. M. Woods

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NAVAL POSTGRADUATE SCHOOL MONTEREY, CALIF. 93940

Evaluation of the Accuracy
of a Lower Confidence Limit Estimate
for Series System Reliability

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Samuel Hall Evans Lieutenant, United States Navy B.S.E.E., University of Texas, 1966

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AL POSTGRADUATE SCHOOL

VTEREY, CALIF. 93940

ABSTRACT

The purpose of this study is to evaluate the accuracy of a procedure used to compute an estimate of the lower $100(1-\gamma)\%$ confidence limit for reliability of a system of independent components connected in logical series. The procedure takes a Bayesian approach and uses test data on the individual components where the sample sizes may be unequal and no knowledge of the component failure distribution is needed. A computer simulation is used to generate test failure data and to compute estimates for the lower $100(1-\gamma)\%$ confidence limit on system reliability.

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I. INTRODUCTION

The problem of determining economically a lower $100(1-\gamma)\%$ confidence limit on the system reliability of complex and expensive systems has plagued reliability managers for a long time. In 1968 Joseph Bram of the Center for Naval Analyses developed an approximate procedure to attack this problem for systems of independent components connected in logical series. It is the purpose of this study to test the accuracy of Bram's procedure and to determine the limits of its usefulness, especially in applications involving systems of ten or more components. The accuracy measurements are determined through a computer simulation.

In an effort to demonstrate reliability goals on expensive systems, it is usually not economically feasible to test the entire system many times. Bram takes a Bayesian approach using test data, successes and failures, on individual components or subsystems to obtain an estimate for the lower confidence limit on the overall system reliability. The method assumes that the components are independent, and test sample sizes for the various components may be unequal. No knowledge or assumptions about the components' failure rates is needed for the computation, only the test failure data.

¹ Center for Naval Analyses, O.E.G. Research Contribution No. 79, Confidence Limits for System Reliability, by Joseph Bram, 2 February 1968.



II. RELIABILITY MODEL AND COMPUTATION PROCEDURES

A. THE RELIABILITY MODEL AND COMPUTATION PROCEDURE: METHOD A

1. Summary of the Reliability Model

Consider k independent components, as in Figure 1, connect in logical series with respective individual reliabilities p_1, p_2, \ldots, p_k . By the product rule for series systems, the actual overall system reliability, R_s , is given by

$$R_{s} = \prod_{i=1}^{k} p_{i}.$$

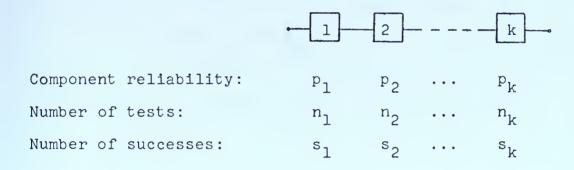


FIGURE 1.

THE SERIES SYSTEM MODEL

In Bram's Bayesian approach to finding an estimate for the lower confidence bound on system reliability, the p_i 's, and hence R_s , are considered to be random variables. If the distribution from which the p_i 's were sampled were known, a lower confidence bound could be calculated without

testing. Since this distribution is not known however, the Bayesian's assume a prior distribution which when combined with the test data leads to a posterior distribution, hopefully one that is recognizable and from which confidence limits can easily be computed.

By considering the random variable

$$u = - \ln R_s = - \sum_{i=1}^{k} \ln p_i$$

the procedure approximates the density of u by

$$\phi(u) = \frac{\beta^{\alpha+1}}{\alpha!} u^{\alpha} e^{-\beta u}$$

The parameters α and β will be estimated from test data. This density results from having chosen a prior density for p_i which is questionable on the grounds that it is not normalizable (see prameterization of the prior density, Ref. 1) and leads to a posterior which is also not normalizable. The choice of priors causes difficulty in estimating the parameters of $\phi(u)$ when no failures occur on any of the components, and its effects will be investigated later in this study. However, since this is an approximate method, the development can proceed from the assumption that u does in fact possess the above density and that estimates for α and β can be determined.

If a new random variable v is defined where

 $v = 2\beta u$,



then v is distributed "chi square" with $2(\alpha+1)$ degrees of freedom. Then the lower $100(1-\gamma)\%$ confidence limit, $R_{s,L(\gamma)}$, can be found as follows:

$$P[2\beta u < \chi^{2}_{2(\alpha+1),1-\gamma}] = 1 - \gamma$$

$$P[u < \frac{\chi^{2}_{2(\alpha+1),1-\gamma}}{2\beta}] = 1 - \gamma$$

$$P[e^{-u} > \exp(-\frac{\chi^{2}_{2(\alpha+1),1-\gamma}}{2\beta})] = 1 - \gamma$$

$$P[R_{s} > \exp(-\frac{\chi^{2}_{2(\alpha+1),1-\gamma}}{2\beta})] = 1 - \gamma$$

From the probability statement above,

$$R_{s,L(\gamma)} = \exp\left(-\frac{\chi_{2(\alpha+1),1-\gamma}^2}{2\beta}\right)$$

is the desired approximate lower confidence limit. The problem is to find an estimate, $\hat{R}_{s,L(\gamma)}$, for $R_{s,L(\gamma)}$ which will yield values sufficiently accurate for the reliability manager's purposes. Bram's method specifies estimates $\hat{\alpha}$ and $\hat{\beta}$, which are computed from the test data, and substituted into the above expression, to obtain

$$\hat{R}_{s,L(\gamma)} = \exp\left(-\frac{\chi_{2(\alpha+1),1-\gamma}^2}{2\hat{\beta}}\right).$$

In the following section Bram's computation procedure for $\hat{\alpha}$ and $\hat{\beta}$ is described.



2. Description of the Computation Procedure

In the following, Bram's procedure for computing $\hat{R}_{s,L(\gamma)}$, will be referred to as method A, and the author's modifications to the original procedure will be referred to as methods B and C.

For descriptive purposes, after testing has been done, consider the k system components to be arranged as follows: 1, 2,..., k_1 , k_1 +1,..., k, where components 1, ..., k_1 are those which experience no failures, and k_1 +1, ..., k are those which experience one or more failures. Let n_i be the number of tests on component i and s_i be the number of successes in the n_i tests. To get \hat{R}_s , $L(\gamma)$ by method A proceed as follows:

$$\hat{M}_{A} = \begin{array}{c} k & n_{1}-1 \\ \Sigma & \Sigma \\ i=k_{1}+1 & j=s_{1} \end{array},$$

$$\hat{v}_{A} = \begin{bmatrix} k & n_{i}-1 \\ \Sigma & \Sigma \\ i=k_{1}+1 & j=s_{i} \end{bmatrix} \frac{1}{j^{2}} ,$$

$$\hat{\alpha} = \frac{\hat{M}_A^2}{\hat{V}_A} - 1$$
 and $\hat{\beta} = \frac{\hat{M}_A}{\hat{V}_A}$, and

$$\hat{R}_{s,L(\gamma)} = \exp\left(-\frac{\chi_{2(\alpha+1),1-\gamma}^2}{2\hat{\beta}}\right)$$

 \hat{M}_A and \hat{V}_A are estimates of the mean and variance of u under method A. In cases where $(\hat{\alpha}+1)$ is non-integer, interpolation is to be used.



3. Shortcomings of Method A

It should be noted that the double sums do not account for components which experience no failures in testing, and if none of the system's components fails, $\hat{\alpha}$ and $\hat{\beta}$ are not defined: under these circumstances the procedure can not be used as it stands. Herein lies the major obstacle to application of the procedure to highly reliable system testing programs where it is quite possible that no failures will occur. In testing the accuracy of method A, whenever no failures occur, $\hat{R}_{s,L(\gamma)}$ is set equal to unity by the author's choice, since Bram makes no allowance for this possibility.

Secondly, as $\hat{\alpha}$ is defined, it is possible that $2(\hat{\alpha}+1)$, the estimate of the degrees of freedom of the chi square random variable can be less than one. Since a chi square variate must have at least one degree of freedom, some provision must be made to account for this inconsistency. For computation purposes in testing the original procedure of method A and the modifications, whenever $2(\hat{\alpha}+1)$ is less than one, the degrees of freedom are set equal to one by the author's choice. For all fractional values of $2(\hat{\alpha}+1)$ greater than one, linear interpolation between the tabulated integer degrees of freedom are used, as specified in the original procedure.

B. FIRST MODIFICATION TO THE COMPUTATION PROCEDURE: METHOD B

The only way that the true lower confidence limit R s,L(γ) can be unity is to have no failures on an infinite number of



tests; this would be impossible to demonstrate. In searching for a more realistic value for $\hat{R}_{s,L(\gamma)}$ when no failures occur, one might be led to believe that $\hat{R}_{s,L(\gamma)}$ should be close to but not equal to one. Due to its discrete character, the procedure of method A will compute one value of $\hat{R}_{s,L(\gamma)}$ which is closest to unity; this will happen when one failure occurs on the component with the greatest number of tests. An approximation of a more accurate estimate of $\hat{R}_{s,L(\gamma)}$ when no failures occur, can be obtained by using one half the original estimates \hat{M}_A and \hat{V}_A , where one failure occurs on the component with the most tests. This has the effect of considering a partial failure on the component with the most tests and tends to smooth the discrete distribution of values of $\hat{R}_{s,L(\gamma)}$ near unity.

In summary, computations under method B proceed as follows:

$$\hat{\mathbf{M}}_{\mathrm{B}} = \hat{\mathbf{M}}_{\mathrm{A}}$$

$$\hat{\mathbf{V}}_{\mathrm{B}} = \hat{\mathbf{V}}_{\mathrm{A}}$$
 if $\mathbf{k}_{\mathrm{l}} < \mathbf{k}$, and

$$\hat{N}_{B} = \frac{1}{2(N_{\text{max}} - 1)}$$

$$\hat{V}_{B} = \frac{1}{2(N_{\text{max}} - 1)^{2}}$$
if $k_{1} = k$.

 $N_{\mbox{max}}$ is the number of tests on the component with the greatest number of tests.



$$\hat{\alpha} = \frac{\hat{M}_B^2}{\hat{V}_B} - 1 \quad ,$$

$$\hat{\beta} = \frac{\hat{M}_B}{\hat{V}_B}$$
 , and

$$\hat{R}_{s,L(\gamma)} = \exp[-\frac{\chi_{2(\alpha+1),1-\gamma)}^2}{2\hat{B}}]$$

If $2(\hat{\alpha}+1) < 1.0$, then set $2(\hat{\alpha}+1) = 1.0$.

C. SECOND MODIFICATION TO THE COMPUTATION PROCEDURE: METHOD C

In an effort to refine method B so that it might be useful under conditions of very high component reliability and small amounts of testing, a second modification is presented. Method C of computing $\hat{\mathbb{N}}$ and $\hat{\mathbb{V}}$ is an attempt to further smooth the distribution of $\hat{\mathbb{N}}_{s,L(\gamma)}$ and has the effect of adding a partial failure to each component which experiences no failure during testing. For method C computation proceeds as follows:

$$\hat{M}_{C} = \sum_{i=1}^{k_{1}} \frac{1}{2 k_{1}^{2} (n_{i} - 1)} + \hat{M}_{A}$$

$$\hat{V}_{C} = \sum_{i=1}^{k_{1}} \frac{1}{2 k_{1}^{2} (n_{i} - 1)^{2}} + \hat{V}_{A}$$



$$\hat{\alpha} = \frac{\hat{M}_C^2}{\hat{V}_C} - 1 \quad ,$$

$$\hat{\beta} = \frac{\hat{M}_{C}}{\hat{V}_{C}}$$
 , and

$$\hat{R}_{s,L(\gamma)} = \exp\left[-\frac{\chi_{2(\alpha+1),1-\gamma}^2}{2\hat{\beta}}\right]$$

Again, if $2(\hat{\alpha}+1) < 1.0$, set $2(\hat{\alpha}+1) = 1.0$.

From these computations it is seen that if every component experiences at least one failure, i.e. k_1 = 0, then method C reduces to method B. On the other hand, if a system consists of say forty components, and there is limited testing relative to the component unreliability of say ten of the components, one might feel that some adjustment should be made to the overall estimate $\hat{R}_{s,L(\gamma)}$ to account for the lack of testing on those ten components; that is the motivation for this second modification. Furthermore, any method of computing $\hat{R}_{s,L(\gamma)}$ which yields accurate results with less testing than another method will be more desirable economically.

		,

III. SIMULATION PROCEDURE

The purpose of the computer simulation in this study is to demonstrate the accuracy of the procedures described in section II. The simulation considers the system of k components of Figure 1, where the component reliabilities and hence $R_{\rm g}$ are assumed to be known. For this system tests are simulated by the Monte Carlo technique, generating failure data which is used to compute $\hat{R}_{s,L(\gamma)}$ by each of the three methods A, B, and C. By replicating the procedure 1000 times, 1000 values of $\hat{R}_{s,L(\gamma)}$ are generated which are then arranged in ascending order. If the method of computing $R_{s,L(\gamma)}$ is accurate, then $100(1-\gamma)\%$ of the time the true system reliability, R_s , will be greater than or equal to $\hat{R}_{s,L(\gamma)}$. Therefore the $100(1-\gamma)^{th}$ percentile point of the 1000 ordered values of $\hat{R}_{s,L(\gamma)}$, $A_{l-\gamma}$, should be equal to or very close to the value of $R_{_{\mathbf{S}}}$ if the method is accurate. For example, if $\gamma = .10$ then the 90th percentile point or 900th ordered value of $\hat{R}_{s,L(\gamma)}$ should be close to R_s . It is this difference, $|R_s - A_{1-\gamma}|$, that is the primary measure of accuracy of the method of estimation. Another measure of the accuracy is the true level of confidence. The true level of confidence is found by determining the percentage of the values of $\hat{R}_{s,L(\gamma)}$ that are less than or equal to R_s . If the method is accurate the true level of confidence will be very close to $100(1-\gamma)\%$.



One of the main concerns of this study is to determine what minimum amount of testing is necessary to insure that the method will provide accurate results. This minimum amount of testing will be different for each system and will be dependent on the number of components and their reliabilities. For this reason the quantity

$$TT = \sum_{i=1}^{k} n_i (1-p_i)$$

is used as a measure of the amount of testing relative to the system reliability. Low values of TT occur when combinations of high reliability and small numbers of tests occur, and TT increases directly with the number of tests per component and inversely with component reliability. Intuitively one would feel that as TT decreases the accuracy of the computation procedure will decrease since fewer failures will occur, and less information will be obtained. One purpose of the simulation is to determine for what values of TT the various methods meet desired accuracy goals.

²Borsting, Jack R., and Woods, W. Max, <u>A Method for Computing Lower Confidence Limits on System Reliability Using Component Failure Data With Unequal Sample Sizes, Naval Postgraduate School, Monterey, California, June 1968.</u>



IV. RESULTS AND CONCLUSIONS

The analysis of accuracy results are shown in Tables I and II. The cases considered are grouped according to the number of components, system reliability, and value of TT. The component reliabilities, p_i's, are held constant in some cases while the number of tests per component, n_i's, are varied, to yield a range of TT values. In other cases the n_i's are fixed and the p_i's varied. Twenty-three cases of systems with reliabilities ranging from approximately .87 to .95 and consisting of 10, 20, and 40 components are considered.

As expected the tendency of all methods to increase in accuracy as TT increases is substantiated. In addition, the tendency of increased accuracy from methods A, to B, to C is also substantiated. Cases 21, 22, and 23 show clearly that for TT = 10.0 the accuracy of the three methods is essentially the same for all three methods of computation, since with this relatively high amount of testing the methods become identical. These three cases also show that as the number of system components changes, if R_S and TT are held constant, the accuracy of the three methods remains essentially unchanged.

It is up to the user to determine what accuracy is acceptable when employing any procedure to compute $\hat{R}_{s,L(\gamma)}$. For illustrative purposes, if one considers $|R_s-A_{l-\gamma}|<.02$ to be acceptable accuracy, then for a 40 component system

	*	

with an overall reliability of approximately .90, the testing program should be such that TT = 6.50 is a minimum and TT > 9.0 would be desirable. As TT decreases below 6.50 there is a marked deterioration in accuracy of all three methods as measured by both $|R_s - A_{l-v}|$ and the true level of confidence versus the desired level of confidence. Another key factor to consider in using any of these methods is that each is sensitive to the individual component test sample size, n;. If n, is quite small, say less than 10, for any component regardless of TT, a failure on one of those n; tests will tend to have a disproportionate effect on $\hat{R}_{s,L(\gamma)}$. Note that in cases 6 through 10 accuracy proves to be not as good as in cases 1 through 5 for k = 20 and corresponding values of TT. This is because the n, values are generally lower for cases 6 through 10 and in some cases n, is less than 10.

The results show little difference in accuracy between the 80% (γ = .20) and 90% (γ = .10) confidence levels among the various cases and methods. There is no general tendency for the accuracy to be better at either level although one might expect that accuracy at the 80% confidence level would be better than at the 90% level. For example in case 14A (TT = 6.55) accuracy is better at the 90% level, but in cases 13A (TT = 4.05) and 15A (TT = 9.10) accuracy is better at the 80% level.

These methods are approximations. No analytical justification is given for the modifications suggested. Intuition



has been the guiding factor in development of the methods and the accuracy tests which are based on sound analytical techniques are the basis of acceptance or rejection of the method as a useful tool. Since systems vary widely in their composition, it is not possible to cover them all with a set of specific rules as to when it would be appropriate to use one of the methods for computing $\hat{R}_{s,L(\gamma)}$. However, from these results two general rules follow:

- 1. For systems of ten to forty independent components TT greater than 10.0 should yield accuracies of $|R_{\rm S}-A_{\rm 1-\gamma}|<.01~{\rm provided~that},$
- 2. no component is tested less than 10 times, i.e. $n_{\text{i}} \geq 10 \text{ for all components.}$

If these two rules are followed it can also be concluded that method B should be employed since there should be no appreciable difference in the accuracies of methods A, B, and C, and B is easier to compute than C. Also B does not suffer from the possibility, remote as it would be if TT were greater than 10.0, of yielding $\hat{R}_{s,L(\gamma)} = 1.0$ as would method A. Any system under consideration can be tested using using the computer simulation procedure, and as TT decreases below 10.0, either because of economic or time constraints in the testing program, this would be recommended.



CASE NO.	Ж	n _i and p _i	TT	Rs	×	A1-8	Rs-A1-8	TRUE CONF. LEVEL
1 A	20	n1=16, i=1,,5; n1=20, i=6,,10; n1=40, i=11,,20; p1=.9975, i=1,,20.	1.45	.9512	.20	1.0000	.0488	.563
1 B	20	. 11	1.45	.9512	.20	.9659	.0147	NC NC
1 C	20		1.45	.9512	.20	.9402	.0110	1.000 NC
2 A	20	n1=6, i=1,,5; n1=30, i=6,,10; n1=100, i=11,,20. pi=.9975, i=1,,20.	2.95	.9512	.20	.9770	.0258	.514
2 B	20		2.95	.9512	.10	NC NC	NC NC	NC NC
	20		2.95	.9512	.20	.9566	.0054	.563
*NC	}	indicates No Change from corresponding	1	result of pa	previous	method.		

NU INDICATES NO UNANGE TROM CORRESPONDING RESULT OF PREVIOUS METHOD. Summary of Accuracy Results by Case TABLE I.



CASE NO.	X	n _i and p _i	TT	Rs	×	A1-8	Rs-A1-8	TRUE CONF. LEVEL
3A	20	ni=14 , i=1,,5; ni=60 , i=6,,10; ni=125, i=1,,20; pi=.9975, i=1,,20.	4.05	.9512	.20	.9691	.0179	.714
3B	20	11	4.05	.9512	.20	NC	NC NC	NC NC
3 C	20	11	4.05	.9512	.20	.9659	.0147	.756 NC
4 A	20	n;=24 , i=1,,5; n;=100, i=6,,10; n;=200, i=11,,20; p;=.9975, i=1,,20.	6.55	.9512	.20	.9669	.0096	.708
4 B	20		6.55	.9512	.10	NC NC	NC	NC NC
7 C	50		6.55	.9512	.20	.9627	.0088	.765 NC
\ \ \ \ \ \ \ \ \ \ \ \ \ \ \ \ \ \ \	}	indicator No Change from correction		20cm1+ 0f m	3110 : 310 %	#0+hod		

INC indicates No Change from corresponding result of previous method. Summary of Accuracy Results by Case TABLE I.



CASE NO.	×	n; and p;	TT	Rs	b	A1-8	Rs-A1-8	TRUE CONF. LEVEL
5 A	50	n1=28 , 1=1,,5; n1=100, 1=6,,10; n1=300, 1=11,,20; p1=.9975, 1 = 1,,20.	9.10	.9512	.10	.9650	.0138	.622
5 B	20	t	9.10	.9512	.20	NC	NC	N N
5 C	20		9.10	.9512	.20	.9635	.0123	.730
6 A	20	n1=8 , 1=1,,5; n1=10, i=6,,10; n1=20, 1=11,,20; p1=.995, 1=1,,20.	1.45	9406.	.20	1.0000	.0954	.521
6 B	50	E	1.45	9006.	.10	.9312	.0266	NC NC
9 C	20	Ξ	1.45	9406.	20	.8808	.0238	NC NC
*	.,	Section No Other Control of the Cont		1	210;202	m + h o d		

"NC indicates No Change from corresponding result of previous method. Summary of Accuracy Results by Case TABLE I.



CASE NO.	×	n; and p;	TT	Rs	×	A1-8	Rs-A1-8	TRUE CONF. LEVEL
7.A	20	n ₁ =3 , i=1,,5; n ₁ =15, i=6,,10; n ₁ =50, i=11,,20; p ₁ =.995, i=1,,20.	2.95	9406.	.10	.9541	.0360	. 638
7 B	50		2.95	.9046	.10	NC NC	NC NC	N C
7 C	20	Ε	2.95	9406.	.10	. 8946	.0100	NC .719
8 A	20	ni=7, i=1,,5; ni=35, i=6,,10; ni=60, i=11,,20; pi=.995, i=1,,20.	4.05	9406.	.10	.9362	.0316	.575
8 8	20		4.05	9406.	.20	NC NC	NC NC	NC
υ ∞	20		4.05	9406.	.20	.9295	.0249	.713
* N	1	pripactory wort operation of their		result of n	nrewinns	method		

NC indicates No Change from corresponding result of previous method. Summary of Accuracy Results by Case TABLE I.



CASE NO.	×	n; and P;	TT	Rs	مر	A1-8	Rs-A1-8	TRUE CONF LEVEL
9 A	20	n;=12 , i=1,,5; ni=50 , 1=6,,10; ni=100, i=11,,20; pi=.995, i=1,,20.	6.55	9006.	.20	.9346	.0300	.618
9 B	20	11	6.55	9406.	.20	NC	NC NC	N C
) 6	20	12	6.55	9006.	.20	.9312	.0266	.749
10A	20	ni=14 , i=1,,5; ni=50 , i=6,,10; ni=150, i=11,,20; pi=.995, i=1,,20.	9.10	.9046	.20	.9315	.0269	.723
10B	20	н	9.10	9406.	.20	NC NC	NC NC	NC
10 C	20		9.10	9406.	.20	.9283	.0237	.726
₹ *		the My Change from correcting	1	recitle of ny	T. C. L. C.	mothod		

'NC indicates No Change from corresponding result of previous method. Summary of Accuracy Results by Case TABLE I.



CASE	×	n; and p;	TT	Rs	×	A1-8	Rs-A1-8	TRUE CONF.
11A	40	ni=5 , i=1,15; ni=10, i=16,,30; ni=50, i=31,,40; pi=.998, i=1,,40.	1.45	.9230	.10	1.0000	0770.	.419 .410
11 B	40	¥	1.45	.9230	.20	.9727 NC	.0497 NC	NC
11 C	40	Ξ	1.45	.9230	.10	.9450	.0220	.610
12A	40	n1=5 , 1=1,,15; n1=10 , 1=16,,30; n1=100, 1=31,,35; n1=1508,1=36,,40;	2.95	.9230	.20	.9847	.0617	.378
12B	40	Ε	2.95	.9230	.20	NC NC	NC	NC NC
12C	. 40	=	2.95	.9230	.20	.9539	.0309	.495
*NC		indicates No Change from corresponding	1	result of p	previous	method.		

Summary of Accuracy Results by Case TABLE I.



CASE NO.	Ж	n _i and p _i	TT	Rs	Ø	A1-8	Rs-A1-8	TRUE CONF. LEVEL
13A	70	n ₁ =5,i=1,5; n ₁ =10,i=6,,10; n ₁ =20,i=11,,20;n ₁ =50,i=21,,30; n ₁ =100,i=31,,35;n ₁ =150,i=36,,40; p ₁ =.998, i=1,,40.	4.05	.9230	.10	.9650	.0420	.570
13B	40		4.05	.9230	.20	NC NC	O O N N	NC NC
13 C	40	•	4.05	. 9230	.20	.9524	.0294	.594
14 A	0 †7	n:=5,i=1,,5;n;=50,i=6,,20; n:=100, i=21,,30; n:=150, i=31,,40; p:=.998, i=1,,40.	6.55	.9230	.10	.9396	.0198	.785
14 B	40	-	6.55	.9230	.20	N N	NC NC	N N
14 C	0 †7		6.55	.9230	.20	.9356	.0126	.825
»NC		indicates No Change from corresponding		result of ny	nrevious	method.		

NC indicates No Change from corresponding result of previous method. Summary of Accuracy Results by Case TABLE I.



CASE NO.	×	n _i and p _i	TT	Rs	×	A1-8	Rs-A1-8	TRUE CONF. LEVEL
15A	7 0	n1= 50, i=1,,10; n1=100, i=11,,20; n1=150, i=21,,40; pi=.998, i=1,,40.	9.00	.9230	.10	.9294	.0064	.861
15B	0 †7		9.00	.9230	.10	NC NC	NC NC	NC NC
15 C	7 0	11	9.00	.9230	.10	.9292	.0062	. 863 . NC
16 A	40	n ₁ =15, i=1,,40 p ₁ =.999,i=1,,5;p ₁ =.998,i=6,,20 p ₁ =.997, i=21,,35; p ₁ =.990, i=36,,40.	20 1.95	.8778	.10	1.0000	.1222	.851
16B	40		1.95	.8778	.10	.9078 NC	.0300 NC	NC NC
16C	40		1.95	.8778	.10	NC .8902	NC .0124	NC NC
*NC	}	indicates No Change from corresponding		result of p	previous	method.		

Summary of Accuracy Results by Case TABLE I.



CASE NO.	K	n; and p;	TT	R _S	×	A1-8	Rs-A1-8	TRUE CONF. LEVEL
17 A	017	n1=20, 1=1,,40; p1=.999,1=1,,5;p1=.998,1=6,,20; p1=.997, 1=21,,35; p1=.990, 1=36,,40.	0; 2.60	.8778	.20	.8858	.0080	.746
17 B	0 †7	E	2.60	.8778	.20	NC NC	NC NC .	NC
17 C	40	Ξ	2.60	.8778	.20	.8848	.0070	.735
18A	7 0	n ₁ =30, 1=1,,40 p ₁ =.999,1=1,,5;p ₁ =.998,1=6,,20; p ₁ =.997,1=21,,35; p ₁ =.990,1=36,,40	3.90	.8778	.10	.9236	.0458	. 746
18B	40		3.90	.8778	.20	NC	NC NC	NC NC
18C	710		3.90	.8778	.20	.9229	.0451	NC NC
* NC		indicates No Change from corresponding		result of pr	previous	method.		

Summary of Accuracy Results by Case TABLE I.



CASE NO.	×	n _i and p _i	TT	Rs	b	A1-8	Rs-A1-8	TRUE CONF. LEVEL
19 A	7 0	n ₁ =50, i=1,,40 p ₁ =.999,i=1,,5;p ₁ =.998,i=6,,20; p ₁ =.997,i=21,,35; p ₁ =.990,i=36,,40.	6.50	.8778	.10	.8975	.0197	.884
19 B	40	Ε	6.50	.8778	.20	NC NC	NC NC	O O
7.6 C	40	11	6.50	.8778	.20	.8971	.0193	NC
20 A	7,0	n ₁ =70, i=1,,40 p ₁ =.999,i=1,,5;p ₁ =.998,i=6,,20; p ₁ =.997,i=21,,35; p ₁ =.990,i=36,,40.	9.10	.8778	.10	.8745	.0033	.902
20 B	40	11	9.10	.8778	.20	NC NC	NC NC	NC NC
20 C	0 7		9.10	.8778	.20	.8743	.0035	NC
UN *	1	indicates No Change from corresponding		result of pr	nrevious	method		

INC indicates No Change from corresponding result of previous method. Summary of Accuracy Results by Case TABLE I.



CASE	×	n. and n.	TT	R	>	A1 ×	R A1_x	TRUE
NO		-1 £1		S	5	0 _ T		CONFI
21A	10	n ₁ =125, 1=1,,10; p ₁ =.992, i=1,,10.	10.0	.9228	.10	.9279	.0051	.879
21 B	10	. H	10.0	.9228	.20	NC NC	NC	NC
21 C	70	11	10.0	.9228	.20	.9270	.0042	NC
22 A	20	n ₁ =125, i=1,,20; p ₁ =.996, i=1,,20.	10.0	.9229	.20	.9281	.0052	. 793
22 B	20	=	10.0	.9229	.20	NC NC	NC NC	NC NC
22 C	20		10.0	.9229	.20	.9278	.0048	NC NC
NON *	1	indicates No Change from corresponding	1	result of pr	previous	method		

INC indicates No Change from corresponding result of previous method. Summary of Accuracy Results by Case TABLE I.



CASE NO.	×	n _i and p _i	TT	Rs	×	A1-8	Rs-A1-8	TRUE CONF.
23 A	0 †7	n ₁ =125, i=1,,40; p ₁ =.998, i=1,,40.	10.0	.9230	.10	.9281	.0051	.885 .779
23 B	0 †		10.0	.9230	.20	N C	N N	NC NC
23 C	0 †		10.0	.9230	.20	.9280	. 0050	NC
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*NC		indicates No Change from corresponding		result of pa	previous	s method.		

Summary of Accuracy Results by Case TABLE I.



		-		Rs - Al-V		
Case No.	k	TT	R _s	A	В	C
16	40	1.95	.8778	.1222	.0300	NC
6	20	1.45	.9046	.0954	.0266	.0238
11	40	1.45	.9230	.0770	.0497	.0220
1	20	1.45	.9512	.0488	.0147	.0110
17	40	2.60	.8778	.0080	NC	.0070
7	20	2.95	.9046	.0495	NC	.0100
12	40	2.95	.9230	.0617	NC	.0309
2	20	2.95	.9512	.0258	NC	.0054
18	40	3.90	.8778	.0458	NC	.0451
8	20	4.05	.9046	.0316	NC	.0249
13	40	4.05	.9230	.0420	NC	.0294
3	20	4.05	.9512	.0179	ЙС	.0147
19	40	6.50	.8778	.0197	NC	.0193
9	20	6.55	.9046	.0300	NC	.0266
14	40	6.55	.9230	.0166	NC	.0,126
4	20	6.55	.9512	.0157	NC	.0115

Summary of Accuracy Results for %= .10
Table II



Case	k	TT	$\mathtt{R}_{\mathtt{s}}$	R _S - A _{1-γ}		
No.			5	A	Д	C
20	40	9.10	.8778	.0033	NC	.0035
10	20	9.10	.9046	.0269	NC	.0237
15	40	9.00	.9230	.0064	NC	.0062
5	20	9.10	.9512	.0138	NC	.0123
21	10	10.0	.9228	.0051	NC	.0042
22	20	10.0	.9229	.0052	NC	.0048
23	40	10.0	.9230	.0051	NC	.0050
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Summary of Accuracy Results for %=.10
Table II



			_	Rs - Al-Y		
Case No.	k	TT	R _s	A	В	C
16	40	1.95	.8778	.0136	NC	.0124
6	20	1.45	.9046	.0954	.0532	.0117
11	40	1.45	.9230	.0447	NC	.0414
1	20	1.45	.9512	.0488	.0280	.0088
17	40	2.60	.8778	.0080	NC	.0070
7	20	2.95	.9046	.0360	NC	.0243
12	40	2.95	.9230	.0571	NC	.0420
2	20	2.95	.9512	.0190	NC	.0138
18	40	3.90	.8778	.0239	NC	.0234
8	20	4.05	.9046	.0278	NC	.0213
13	40	4.05	.9230	.0318	NC	.0295
3	20	4.05	.9512	.0148	NC	.0129
19	40	6.50	.8778	.0055	NC	.0059
9	20	6.55	.9046	.0246	NC	.0229
14	40	6.55	.9230	.0198	NC	.0151
4	20	6.55	.9512	.0096	NC	.0088

Summary of Accuracy Results for $\mbox{\em γ}=.20$ Table II



				R _S - A _{l-V}		
Case No.	k	TT	R _s	A	В	C
20	40	9.10	.8778	.0014	NC	.0016
10	20	9.10	.9046	.0215	NC	.0181
15	40	9.00	.9230	.0052	NC	.0050
5	20	9.10	.9512	.0115	NC	.0159
21	10	10.0	.9228	.0062	NC	.0052
22	20	10.0	.9229	.0061	NC	.0052
23	40	10.0	.9230	.0062	NC	.0060
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Summary of Accuracy Results for V= .20 Table II



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11. SUPPLEMENTARY NOTES

12. SPONSORING MILITARY ACTIVITY

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13. ABSTRACT

The purpose of this study is to evaluate the accuracy of a procedure used to compute an estimate of the lower $100(1-\gamma)\%$ confidence limit for reliability of a system of independent components connected in logical series. The procedure takes a Bayesian approach and uses test data on the individual components where the sample sizes may be unequal and no knowledge of the component failure distribution is needed. A computer simulation is used to generate test failure data and to compute estimates for the lower $100(1-\gamma)\%$ confidence limit on system reliability.



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